Development of New Thermal Protection Systems Based on Polysiloxane/Silica Composites*

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Today's Presentation

- Background and Motivation
- Research Objective
- Characterization of the Neat Resins
- Sample Manufacturing
- Ablation Testing
- Microstructure Analysis
- Conclusion and Future work



Background

- Want to investigate new ablative material systems
- Need for materials which can withstand harsher environments
 - Rocket Motors
 - TPS Materials for Re-entry Vehicle
 - Nose Cones Atmospheric Probes
 - Vertical Launching Systems
 - Fire Prevention Trains, Submarines, etc.



Ablative Materials

- Resist thermochemical erosion caused by harsh environment
- Formation of the protective char layer
- Different mechanisms for different materials, no one size fits all ablative
- Current SOTA resins: SC-1008, PT-15





Research Objective

- Using legacy material as control, S/Ph
- MX2600 TDS lists 30-35% resin content
- Goal of creating testing various resin content S/DG-UHTR samples
- Find optimal resin:fiber ratio



SC-1008 Phenolic

- MIL-standard phenolic resole resin manufactured by Hexion
- Typically carbon or silica fiber reinforcement
- Foamed versions of phenolic used for low density ablators
- Lots of data collected
- Diverse applications, from TPS materials to rocket motor materials
- Relatively cheap



Silica/Phenolic composite



DG Polysiloxane

- Inorganic matrix, utilizing a mixture of polysiloxane chemistries manufactured Dyna-Glas Technologies LLC
- Pre-ceramic material
- High char yield
- Low heat release rate and heat release capacity
- Good compatibility with silica fabric





Silica-/DG-UHTR composite



Material Characterization

- Thermogravimetric Analysis
 - Thermal Stability & Char Yield
- Microscale Combustion Calorimeter
 - Heat Release Rate and Capacity
- Density
 - Water Displacement
- Oxygen-Acetylene Test Bed
 - Ablative Performance



Char Yield Study

- 1. Dry the TGA sample 150°C for 30min
- 2. Consistent sample size 20mg
- 3. TGA heating rate of 20°C/min in nitrogen
- Char yield is defined as the %mass remaining at 1,000°C

Developed based on a NASA report on PICA







Char yield results for the neat SC-1008 and DG-UHTR resins





Char Yield Study

dTGA for the neat SC-1008 and DG-UHTR resins



Flammability Properties

- Microscale Combustion Calorimeter
- Lab scale for small sizes
- Screening tool
- Good alternative to a cone calorimeter





Flammability Properties





Flammability Properties



Comparison of the Heat Release Capacities for the four resin systems

intern of



Sample Preparation

- Wet lay-up
 - Hand lay-up of fabric with resin
 - Make partially cured pre-preg
- Compression mold
 - 170°C 1hr
 - 250°C 2hr
- Water jet samples
 - 15mm diameter
 - 15-17mm thick
- Drilled holes
 - 10mm from the surface





Test Samples

Sample ID	Resin %	Fiber %	Silica filler %	Nanosilica wt% (in resin)
S/Ph MX2600	30-35% (burn-off)	64%	4.5%	0%
S/DG-UHTR F1	35%	65%	0%	0%
S/DG-UHTR F2	40%	60%	0%	0%
S/DG-UHTR F3	48%	52%	0%	0%
S/DG-UHTR F4	35%	75%	0%	3%



Oxygen-Acetylene Test Bed

- Welding torch setup utilizing high flow rates
- Utilized a 1:1 Oxygen to Fuel ratio
- Tested at a heat flux of 1000W/m², verified using gardon gauge
- 40s exposure time
- Carbon-carbon shield





Experimental Setup

- Test Equipment
 - Two color Pyrometer
 - IR camera
 - HD camera
 - K-type thermocouple





Ablation Testing



15mm diameter x 15mm thick samples



Thermocouple Data



Surface Temperature Data





TEM





1 um



Density





Recession Rate





Heat Soak Temperature





Mass Loss Percent





TGA





IR Video





S/DG-UHTR F2



IR Video

- Growing blue ring around the sample is the radiative heat
- Yellow/Green circle is the physical sample





HD Video





S/DG-UHTR F2



HD Video



100x Top View - Char layer



S/Ph MX2600

S/DG-UHTR F1

S/DG-UHTR F1

1000x Top View - Char Layer







100x Cross Section– Char Layer





S/DG-UHTR F1



1000x Cross Section– Char Layer







100x Cross Section– Virgin





EDX Analysis – DG-UHTR F1 Char





Conclusion

- All Silica/DG-UHTR formulations showed the best thermophysical and ablation properties
- The 35wt% Silica/DG-UHTR F1 composite showed the highest char yield, lowest recession rate, mass loss, and heat soak temperatures
- Better processing method needed to obtain best quality samples



Future Work

- Characterization of mechanical properties
 - Tensile strength and modulus
 - Compression strength and modulus
 - Flexural strength and modulus
- Test lower S/DG Resin:Fiber ratios: 25wt%, 30wt%
- Ablation testing using Inductively Coupled Plasma (ICP) torch
- Better dispersion of nanosilica into the resin
- Test shear and compression strength of char
- Explorer incorporation different additives and nanomaterials



Thanks for attending! Questions?

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